



FLOW BOILING IN MICROGRAVITY: APPLICATION TO COOLING ELECTRONIC COMPONENTS ON SATELLITES

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MOTIVATION AND CONTEXT

Studies of two-phase flows with phase changes motivated by nuclear safety issues (Critical Heat Flux, Reactivity Insertion Accidents)



ESC-B/Vinci Engine

and space applications re-ignition of cryogenic engines of the launchers (quenching, flow boiling)

- thermal control of electronic components in satellites



DESIGN OF A TWO-PHASE LOOP FOR COOLING ELECTRONIC COMPONENT OF A SATELLITE

MATRAS Programme of FNRAE (2009-2013) with THALES ALENIA SPACE



DESIGN OF A TWO-PHASE LOOP FOR COOLING ELECTRONIC COMPONENT OF A SATELLITE



Modelling of the pressure drop and heat transfer in the whole panel in microgravity and ground conditions \rightarrow optimisation of the design 3 tubes of 12mm diameter in parallel (60m) \rightarrow extraction of 3.4 kW

→ development of a 1D model
Project of INP-N7 students
→ Building of an experimental set-up for ground and microgravity experiments



ESA MAP PROJECT MULTISCALE ANALYSIS ON BOILING (2008-....)

Boiling investigation at the bubble scale (experiments, DNS, ..) \rightarrow RUBI experiments on the ISS (2019)

Flow Boiling in tube in microgravity

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FLOW BOILING IN TUBE

• EXPERIMENTAL SETUP

• MEASUREMENT TECHNIQUES

- VOID FRACTION
- WALL SHEAR STRESS
- HEAT TRANSFER COEFFICIENT

• RESULTS:

• CONCLUSIONS

EXPERIMENTAL SETUP

- Designed and built for two-phase flows studies with phase change 0 under microgravity conditions.
- BRASIL: Boiling Regimes in Annular and Slug flow In Low 0 gravity
- G=50-300 kg/m²/s 0
- x=0-0.6
- ΔT (subcooled) <10°C 0
- \circ q= 0.5-4 W/cm²
- ID=6 mm0
- Fluids HFE7000 HFE7100 0





EXPERIMENTAL SETUP



TEST SECTION: SAPPHIRE TUBE



Sapphire tube:

- 200 mm long sapphire tube
- Semi-transparent with an ITO coating for Joule effect heating
- Wall temperature measured by Pt100 probes
- HFE 7000 ->T_{sat}=34°C @ 1 bar
- Pressure drop measurements →wall shear stress
- Capacitance probes \rightarrow void fraction
- Thermocouples \rightarrow liquid enthalpy and quality x

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copper electrodes



RESULTS: FLOW REGIMES

 $G=200 kg/m^2/s$ $G=50 kg/m^2/s$ $2 W/cm^2$ 2 W/cm² $\Delta T=10^{\circ}C$ $\Delta T=10^{\circ}C$

4 W/cm² T_{saturation}



Microgravity

G=200kg/m²/s G=200kg/m²/s G=50kg/m²/s 2 W/cm² 2 W/cm² $\Delta T=10^{\circ}C$ $\Delta T=10^{\circ}C$

G=200kg/m²/s

4 W/cm²

T_{saturation}

Influence of gravity:

- Size and shapes of bubbles
- At detachment larger bubble diameter is observed in microgravity.
- Liquid film in annular flow seems smoother

Normal gravity



TEST SECTION: SILICON TUBE



Silicon:

- 100 mm long sapphire tube
- Transparent to the IR camera doped for Joule effect heating
- Wall temperature measured obtained from IR camera visualization
- HFE 7100 ->T_{sat}=61°C @ 1 bar
- ESA's High Resolution IR camera



HEAT TRANSFER COEFFICIENT IN SLUG FLOW



RESULTS: VOID FRACTION NORMAL VS MICROGRAVITY



Evolution of the void fraction as a function of mass flow rate (kg/m²/s)

Influence of gravity: - The liquid film is thicker in 1g.





INTERFACIAL FRICTION FACTOR FOR ANNULAR FLOW



INTERFACIAL FRICTION FACTOR FOR ANNULAR FLOW



HEAT TRANSFER COEFFICIENT



Deterioration of the heat transfer in microgravity.

- At lower quality the influence of gravity can be seen,
- At higher quality good agreement with classical correlations.

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CONCLUSION

- Study of flow boiling in tube → expertise in specific measurements technics (Thermocouples, Infrared camera, capacitance probes, pressure drops, high-speed video recording and image processing) and in 1 dimensional modelling of two-phase flows.
- Close connexions with industrial partners:
 Thales Alena Space for the design of two-phase loop for cooling electronic devices
 - Air Liquide : heat and mass transfers in space launchers
 - Snecma moteurs & CNES: chill down of tubes before the re-ignition of space launcher engine (Ariane V programme)
 - IRSN: rapid transient boiling in nuclear reactors (RIA)



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THANK YOU FOR YOUR ATTENTION

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